CRITICAL ITEMS LIST

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REFERENCE DESIGNATOR: F1 NAME/QUANTITY: FUSE/ I

PROJECT: IFM BREAKOUT BOX LRU NAME / QUANTITY: IFM BREAKOUT BOX / 2 SUBSYSTEM: NONE EFFECTIVITY: All Orbiters

ORAWING REFERENCE: 10120-20022

LRU PART NUMBER: SED39121772

FAILURE MODE NE 4	JMBER	CRITICALITY 2/18	FAILURE EFFECT	RETENTION RATIONALE
FUNCTION Fuse outlet 1 provides fuse protection for the IFM breakout box (outlet 1) and downstream equipment			ENDITEM No power output at outlet 1; loss of redundant power to CWE	A. DESIGN – The part is a minia- ture cartridge fuse with leads. It is rated at S.A., with a capacity to interrupt 1,000 A at 60 V, and is not to exceed 3 g. The current load used to power the C&W System is nomi- nally .7 A.
FAILURE MODE AND CAUSE Mode: Fuse opens prematurely Cause: Defective material			MISSION None CREW / VEHICLE This failure followed by failure of the remaining Orbiter essential bus powering the CWE would	B. TESTS – The part is screened and qualified to the requirements of Rockwell International specification MC451-0010. Tests and inspections are done on the entire product to check burn-in (100 percent rating, 2-hr minimum), terminal strength (2 in-lb, 1 min) examination of product vibration (sinusoida) sweep), leakage, do resistance, and radiographics. Tests and inspections
A - Pass	REMARKING PATHS Replace fuse Use backup IFM box		create an undetected fuel cell emergency due to loss of fuel cell coolant pump	are done on a sample from each lot to check terminal strength, vibration (random), leakage, do resistance, radio- graphics, and time current characteristics. The tests and
MISSION PHASE Orbit/Landing	TIME TO EFFECT Minutes	TIME TO CORRECT	See "End Item" and "Crew/Vehicle"	inspections done on a periodic basis for qualification include dc resistance, case leakage, time current characteristics,

PREPARED BY Luis Varquer

REVISION Basic

SUPERSEDING DATE: 8/91

DATE: 8/91

CRITICAL ITEMS LIST

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REFERENCE DESIGNATOR: F1
NAME/QUANTITY: FUSE/1
DRAWING REFERENCE: 10120-24022

PROJECT: IFM BREAKOUT BOX
LRU NAME / QUANTITY: IFM BREAKOUT BOX/2

SUBSYSTEM: NONE EFFECTIVITY: All Orbiters

LRU PART NUMBER: SED39121772

RETENTION RATIONALE (Concluded)

terminal strength, thermal shock, humidity, interrupting capacity, mechanical shock, and vibration. A visual and mechanical examination is also performed.

- C. INSPECTION The part is inspected according to the requirements of Rockwell International specification MC451-0010, which includes visual inspections and burn-in and screening tests as described in item 2. In addition, Rockwell International periodically audits the device manufacturer to ensure that the design, processing assembly, inspection, and testing of devices are adequately controlled.
- D. FAILURE HISTORY None. There have not been any documented failures of a fuse to function on the Orbiter program.
- E. OPERATIONAL USE -
 - Fuse failure would be annunciated in Orbiter failure scenario 2. The fuse failure may be detected in Orbiter failure scenario 1
 by indicator lights on the IFM breakout box. The fuse could then be replaced.
 - 2. The second failure, loss of the Orbiter essential bus, would be detected by the ground except during LOS. There would be 5-10 min (9 min nominal) available to shut down the affected fuel cell.

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REVISION: Sasic

SUPERSEDING DATE: 8/91

DATE: 8/91